

Cessna 208 Series Aeroplanes

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**AD/CESSNA 208/10  
Amdt 2**

**Flap Bellcrank Bolt**

**13/95**

Applicability: Models 208 and 208A S/nos 20800001 to 20800243, and 208B S/nos 208B0001 to 208B0465.

Requirement: Action in accordance with Cessna Service Bulletin CAB95-11 Rev 1.

Compliance: 1. For aircraft where the bellcrank bolt has achieved 500 landings, within 100 hours time in service after 12 October 1995 or prior to 12 October 1996, whichever occurs first.

2. For aircraft where the bellcrank bolt has not achieved 500 landings, prior to achieving 501 landings or within 100 hours time in service after 12 October 1995, whichever occurs last.

3. For aircraft that have been modified by the installation of SK208-123-1 or SK208-123A1, within the next 200 hours time in service after 7 December 1995.

The effective dates of this Directive for:

- the initial issue, was 13 October 1994, and
- for Amendment 1, 12 October 1995.

For Amendment 2, Compliance statements 1. and 2. remain unchanged.

Background: The manufacturer had received reports of failure of the subject bolt, a condition that would result in asymmetric flap.

Amendment 1 is the result of the cancellation of Cessna Service Bulletins CAB89-32 Rev 2 and CAB94-14 Rev 1, the Requirement documents for the initial issue of this Directive, and the issue of the current Requirement document.

Amendment 2, through revision of the Requirement document, now includes aircraft previously modified by SK208-123-1 and SK208-123A1.