



## Airworthiness Directive

**AD No.:** 2019-0232

**Issued:** 16 September 2019

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

### Design Approval Holder's Name:

AIRBUS

### Type/Model designation(s):

A319, A320 and A321 aeroplanes

**Effective Date:** 30 September 2019

**TCDS Number(s):** EASA.A.064

**Foreign AD:** Not applicable

**Supersedure:** None

## ATA 29 – Hydraulic Power – Hydraulic Reservoir Pressurisation Lines – Modification

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### Manufacturer(s):

Airbus, formerly Airbus Industrie

### Applicability:

Airbus A319-131, A319-132, A319-133, A320-231, A320-232, A320-233, A321-131, A321-231 and A321-232 aeroplanes, all manufacturer serial numbers.

### Definitions:

For the purpose of this AD, the following definitions apply:

**The SB:** Airbus Service Bulletin (SB) A320-29-1179.

**Affected part:** Bleed air hoses, having Part Number (P/N) D0001037900000.

**Modification kit:** Bleed air hoses, having P/N 1298481, and elbow, having P/N NAS1761J0406.

### Groups:

Group 1 aeroplanes are those that have an affected part installed.

Group 2 aeroplanes are those that do not have an affected part installed.



An aeroplane having Airbus modification (mod) 160901 embodied in production is a Group 2 aeroplane, provided no affected part has been installed on that aeroplane since its date of manufacture.

**Aeroplane date of manufacture:** The date of transfer of title (ownership) which is referenced in Airbus documentation at the time of first delivery to an operator.

**Reason:**

An occurrence was reported of rupture of a hydraulic reservoir air pressurization hose on an in-service aeroplane, leading to air leakage which was undetectable in normal operation, until subsequent zonal inspection.

This condition, if not corrected, could lead to exposure of the wing structure to high temperatures (possibly above 200°C), possibly resulting in reduced structural integrity of the aeroplane.

To address this potential unsafe condition, Airbus developed mod 160901, introducing bleed air hoses with an additional restrictor, limiting the mass flow in case of rupture of the hose, and published the SB providing instructions for retrofit installation.

For the reasons described above, this AD requires a modification of the hydraulic pressurization lines, replacing the affected part with a modification kit, as defined in this AD. This AD also prohibits (re)installation of affected parts.

**Required Action(s) and Compliance Time(s):**

Required as indicated, unless accomplished previously:

**Modification:**

(1) For Group 1 aeroplanes: Within 66 months after the effective date of this AD, modify the aeroplane by replacing the affected part with a modification kit in accordance with the instructions of the SB.

**Parts Installation:**

(2) Do not install an affected part on any aeroplane, as required by paragraph (2.1) or (2.2) of this AD, as applicable.

(2.1) For Group 1 aeroplanes: After modification of the aeroplane as required by paragraph (1) of this AD.

(2.2) For Group 2 aeroplanes: From the effective date of this AD.

**Ref. Publications:**

Airbus SB A320-29-1179 original issue dated 28 May 2019.

The use of later approved revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.



**Remarks:**

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. This AD was posted on 14 August 2019 as PAD 19-156 for consultation until 11 September 2019. The Comment Response Documents can be found in the [EASA Safety Publications Tool](#), in the compressed (zipped) file attached to the record for this AD.
3. Enquiries regarding this AD should be referred to the EASA Programming and Continued Airworthiness Information Section, Certification Directorate. E-mail: [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu).
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#).
5. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS – Airworthiness Office – EIAS; Fax +33 5 61 93 44 51; E-mail: [account.airworth-eas@airbus.com](mailto:account.airworth-eas@airbus.com).

