



Airworthiness Directive

AD No.: **2023-0073R2**

Issued: **26 November 2025**

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I Part M.A.301, or Annex Vb Part ML.A.301, as applicable, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I Part M.A.303, or Annex Vb Part ML.A.303, as applicable] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

Design Approval Holder's Name:

SAAB AB

Type/Model designation(s):

SAAB 340 aeroplanes

Effective Date: Revision 2: 03 December 2025

Revision 1: 18 July 2023

Original Issue: 19 April 2023

TCDS Number(s): EASA.A.068

Foreign AD: Not applicable

Revision: This AD revises EASA AD 2023-0073R1 dated 11 July 2023.

ATA 24 – Electrical Power – DC Starter Generator – Inspection

Manufacturer(s):

SAAB AB (SAAB), formerly Saab Aircraft AB

Applicability:

SAAB SF340A and SAAB 340B aeroplanes, all serial numbers.

Definitions:

For the purpose of this AD, the following definitions apply:

The SB: SAAB Service Bulletin (SB) 340-24-056, which includes reference to (and a copy of) SAFRAN Power USA, LLC Service Information Letter (SIL) 23080-03X-24-03.

Affected part: DC starter generators, having Part Number (P/N) 23080-031 Mod M, with brushes P/N 23080-2017 installed which were manufactured between 2021 (inclusive) and November 2023 (having a date code 11-23) (inclusive).

Serviceable part: An affected part that is new (not previously installed), or that has been repaired or overhauled (new brushes installed) in accordance with the instructions of SAFRAN Power USA, LLC



SIL 23080-03X-24-03, or that has passed an inspection (all brushes are determined to be approved for continued installation) in accordance with the instructions of the SB.

Improved part: DC starter generators, having P/N 23080-031 Mod M, with brushes P/N 23080-2017 installed which were manufactured before 2021 or after November 2023 (having a date code 12-23 or later).

Reason:

Occurrences have been reported of finding excessive wear on certain brushes installed in affected parts. Particularly the brushes manufactured during 2021 and later show wear that exceeds the expected level during its defined in-service time. Investigation of these events is on-going.

This condition, if not detected and corrected, could lead to starter generator failure(s), possibly resulting in total loss of generated electrical power with consequent reduced functional capability of the aeroplane.

To address this potential unsafe condition, SAAB issued the SB, as defined in this AD, to provide inspection instructions.

For the reason described above, EASA issued AD 2023-0073 (later revised) to require repetitive inspections of each affected part and, depending on findings, replacement with a serviceable part. That AD also required implementation of affected part installation requirements.

After AD 2023-0073R1 was issued Safran Power USA, manufacturer of the affected brushes, implemented an improved tamping process resulting in positive performance improvement.

This AD is revised to confirm that the improved parts are no longer affected by the unsafe condition addressed by this AD.

Required Action(s) and Compliance Time(s):

Required as indicated, unless accomplished previously:

Inspection(s):

- (1) Before exceeding 150 flight hours (FH) since first installation of an affected part on an aeroplane, or since latest replacement of the brushes on an affected part, as applicable, or within 50 FH after 19 April 2023 [the effective date of this AD at original issue], whichever occurs later, and, thereafter, at intervals not exceeding the values as specified in Figure 1 of the SB, as applicable, inspect the affected part in accordance with the instructions of the SB.

Corrective Action(s):

- (2) If, during any inspection as required by paragraph (1) of this AD, the brush hours life remaining of one or more brushes is determined to be not approved for continued installation, as defined in the SB, before next flight, replace the affected part with a serviceable or improved part in accordance with the instructions of the SB.



Terminating Action:

(3) Installation on an aeroplane of an improved part constitutes terminating action for the repetitive inspections as required by paragraph (1) of this AD for that part, provided that the part remains in configuration corresponding to improved part.

Part Installation:

(4) From 19 April 2023 [the effective date of the original issue of this AD], it is allowed to install on any aeroplane an affected part, provided it is a serviceable part, as defined in this AD, and that, following installation, it is inspected as required by this AD.

Reporting Requirements:

(5) Within 30 days after each inspection as required by paragraph (1) of this AD, report the results to SAAB AB. This can be done in accordance with the instructions of the SB.

Ref. Publications:

SAAB SB 340-24-056 original issue dated 31 March 2023, or Revision 1 dated 14 April 2023, or Revision 02 dated 28 June 2023, or Revision 3 dated 07 May 2024 or Revision 4 dated 25 November 2025.

SAFRAN Power USA, LLC SIL 23080-03X-24-03 original issue dated 29 March 2023, or Revision 1 dated 10 April 2023.

The use of later approved revisions of the above-mentioned documents is acceptable for compliance with the requirements of this AD.

Remarks:

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. Based on the required actions and the compliance time, EASA have decided to issue a Final AD with Request for Comments, postponing the public consultation process until after publication.
3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.
5. For any question concerning the technical content of the requirements in this AD, please contact: SAAB AB, SE-581 88 Linköping, Sweden, E-mail: Saab340.techsupport@saabgroup.com.

