



Airworthiness Directive

AD No.: 2026-0005

Issued: 13 January 2026

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I Part M.A.301, or Annex Vb Part ML.A.301, as applicable, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I Part M.A.303, or Annex Vb Part ML.A.303, as applicable] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

Design Approval Holder's Name:

SAAB AB

Type/Model designation(s):

SAAB SF340A and 340B aeroplanes

Effective Date: 27 January 2026

TCDS Number(s): EASA.A.068

Foreign AD: Not applicable

Supersedure: This AD supersedes EASA AD 2024-0042 dated 13 February 2024.

ATA 05 – Time Limits / Maintenance Checks – Airworthiness Limitations Section – Amendment

Manufacturer(s):

Saab AB, formerly Saab AB – Support and Services, Saab Aeronautics, SAAB Aerosystems

Applicability:

SAAB SF340A and 340B aeroplanes, all serial numbers.

Definitions:

For the purpose of this AD, the following definitions apply:

The ALM: SAAB SF340A and 340B Airworthiness Limitation Manual (ALM) Revision 3.

The AMP: The Aircraft Maintenance Programme (AMP) contains the tasks on the basis of which the scheduled maintenance is conducted to ensure the continuing airworthiness of each operated aircraft. For aircraft operated under EU regulations, the operator or the owner ensures compliance with the AMP as stipulated in Commission Regulation (EU) [1321/2014](#).



New and/or more restrictive tasks and limitations: This includes all tasks and limitations that are new and all tasks and limitations for which a threshold or interval was reduced, which were introduced into the ALS, as defined in this AD, since the previous ALS revision that is currently incorporated in the AMP.

Reason:

The airworthiness limitations and/or certification maintenance instructions (CMR) for SAAB SF340A and 340B aeroplanes, which are approved by EASA, are currently defined and published in the SAAB SF340A and 340B ALM document. These airworthiness limitations and/or CMRs have been identified as mandatory for continued airworthiness.

Failure to accomplish these instructions could result in an unsafe condition.

Previously EASA issued EASA AD 2024-0042 requiring the actions described in SAAB SF340A and 340B ALM Revision 2.

Since that AD was issued, Saab AB issued the ALM, as defined in this AD, containing new life limitations.

For the reason described above, this AD retains the requirements of EASA AD 2024-0042, which is superseded, and requires accomplishment of the actions specified in the ALM.

Required Action(s) and Compliance Time(s):

Required as indicated by this AD, unless the actions required by this AD have been already accomplished:

Maintenance Tasks and Replacement of Life Limited Parts:

- (1) From the effective date of this AD, accomplish the following actions, as specified in the ALM, as applicable to aeroplane model and depending on aeroplane configuration:
 - (1.1) Replace each component before exceeding the applicable life limit, and
 - (1.2) Within the thresholds and intervals, accomplish all applicable maintenance tasks.

Corrective Action(s):

- (2) In case of finding discrepancies (as defined in the ALM) during accomplishment of any task as required by paragraph (1) of this AD, within the compliance time specified in the ALM, accomplish the applicable corrective action(s) in accordance with the applicable Saab AB maintenance documentation. If no compliance time is identified in the ALM, accomplish the applicable corrective action(s) before next flight. If a detected discrepancy is not identified in the ALM, before next flight, contact Saab AB for approved instructions and accomplish those instructions accordingly.

AMP Revision:

- (3) Within 12 months after the effective date of this AD, revise the approved AMP by incorporating the new life limitations and associated thresholds described in the ALM, as applicable to aeroplane model and depending on aeroplane configuration.



Credit:

- (4) If, before the effective date of this AD, the AMP has been revised to incorporate the maintenance tasks and life limitations as specified in SAAB SF340A and 340B ALM Revision 2, that action ensures the continued accomplishment of those tasks and limitations.

Consequently, for an aeroplane to which that AMP applies, it is acceptable to accomplish the new and/or more restrictive tasks and limitations as specified in the ALM, as applicable to aeroplane model and depending on aeroplane configuration, within the compliance times as specified in the ALM to comply with paragraph (1) of this AD.

For that AMP, it is acceptable to incorporate the new and/or more restrictive tasks and limitations as specified in the ALM, as applicable to aeroplane model and depending on aeroplane configuration, into the AMP to comply with paragraph (3) of this AD.

Recording AD Compliance:

- (5) When the AMP of an aeroplane has been revised as required by paragraph (3) or (4) of this AD, as applicable, that action ensures continued accomplishment of the tasks and limitations as required by paragraphs (1) and (2) of this AD for that aeroplane. Consequently, after revising the AMP, as required by paragraph (3) or (4) of this AD, as applicable, it is not necessary that accomplishment of individual action is recorded for demonstration of AD compliance on a continued basis.

Ref. Publications:

SAAB SF340A and 340B ALM Revision 3 dated 01 December 2025.

The use of later approved revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

Remarks:

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. This AD was posted on 01 December 2025 as PAD 25-179 for consultation until 05 January 2026. No comments were received during the consultation period.
3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.



5. For any question concerning the technical content of the requirements in this AD, please contact: Saab AB, SE-581 88 Linköping, Sweden,
E-mail: saab340.techsupport@saabgroup.com.

