### EASA

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# **EMERGENCY AIRWORTHINESS DIRECTIVE**

### AD No.: 2012-0098-E

### Date: 04 June 2012

Note: This Emergency Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].

## **Design Approval Holder's Name:** EUROCOPTER

Type/Model designation(s):

SA 365, AS 365 and EC 155 helicopters

TCDS Number: DGAC TCDS No. 159

Foreign AD: Not applicable

Supersedure: None

| ΑΤΑ              | Rotorcraft Flight Manual – Emergency Procedures – Rush Revision   |
|------------------|---|
|                  |   |
| Manufacturer(s): | Eurocopter (formerly EUROCOPTER France, Aerospatiale)   |
| Applicability:   | SA 365N, SA365N1, AS365N2, AS365N3, EC155B and EC155B1 helicopters, all serial numbers.   |
| Reason:          | A serious incident has been reported on an AS 332 helicopter that encountered<br>a hydraulic failure while landing on an offshore oil rig. After extension of the<br>landing gear, a cockpit amber alarm illuminated, indicating a low level hydraulic<br>fluid of the left hand side (LH) hydraulic system.  |
|                  | After landing, the helicopter started to roll towards the edge of the helideck as<br>the flight crew was not aware that wheel brake capability was affected and<br>relied only on the effectiveness of the helicopter's ancillary hydraulic<br>accumulator. The results of the following investigation showed that this alarm<br>was due to hydraulic fluid leakage, caused by the failure of one hydraulic union<br>connector in the LH hydraulic circuit. |
|                  | This condition, if not detected and correctly assessed by the flight crew, could result in reduced control of the helicopter after landing.   |
|                  | EASA issued AD 2012-0059-E to address this potential unsafe condition on<br>Eurocopter SA 330, AS 332 and EC 225 helicopters, requiring a rotorcraft flight<br>manual (RFM) revision to cover any other cause of leakage from the<br>emergency/auxiliary hydraulic circuit that could result in an empty accumulator.   |
|                  | The helicopters to which this AD applies are also fitted with an emergency accumulator (as an option on the SA365N version and in basic configuration for the SA365N1, AS365N2, N3 and EC155) and a potential leak could, in certain operational conditions, lead to the same consequences identified in EASA AD  |

|                            | 2012-0059-E.  |   |
|----------------------------|---|---|
|                            | For the reasons described above, t<br>RFM emergency procedures section<br>applicable RFM.   | this Emergency AD requires amending the on by inserting a Rush Revision (RR) into the   |
| Effective Date:            | 05 June 2012  |   |
| Required Action(s)         | Required as indicated, unless acco  | omplished previously:   |
| and Compliance<br>Time(s): | (1) Within 10 flight hours, or before next flight to a helideck or elevated<br>heliport, whichever occurs first after the effective date of this AD, a<br>the RFM, Section emergency procedures, by inserting into the RFI<br>RR specified in Table 1 of this AD, as applicable to the helicopter r<br>and notify all flight crews accordingly. |   |
|                            | Table 1 - RF  | M Rush Revisions (RR)   |
|                            | Helicopter Model  | RFM RR Number (date code)   |
|                            | SA 365 N  | RR.14A (12-14)  |
|                            | SA 365 N1   | RR.12A (12-14)  |
|                            | AS 365 N2   | RR.8A (12-14)   |
|                            | AS 365 N3   | RR.18A (12-14)  |
|                            | AS 365 N3 RCD (helicopter<br>not fitted with autopilot<br>APM2010)  | rs RCD (12-14)  |
|                            | EC 155 B  | RR.17A (12-14)  |
|                            | EC 155 B1   | RR.12A (12-14)  |
|                            | (2) Revising the RFM with an app<br>incorporates the RR specified<br>helicopter model, is an accepta<br>of paragraph (1) of this AD.  | roved RFM Normal Revision which<br>in Table 1 of this AD, as applicable to the<br>able method to comply with the requirements |
| Ref. Publications:         | Eurocopter RFM RR, as applicable  | to the helicopter model:  |
|                            | SA365N RFM RR.14A (date code  | 12-14)  |
|                            | SA365N1 RFM RR.12A (date code   | 9 12-14)  |
|                            | AS365N2 RFM RR.8A (date code  | 12-14)  |
|                            | AS365N3 RFM RR.18A (date code   | 9 12-14)  |
|                            | AS365N3 RCD, RFM RCD (date co   | ode 12-14)  |
|                            | EC155B RFM RR.17A (date code  | 12-14)  |
|                            | EC155B1 RFM RR.12A (date code   | 9 12-14)  |
|                            | The use of later approved revisions compliance with the requirements of   | s of these documents is acceptable for<br>of this AD.   |
| Remarks:                   | 1. If requested and appropriately<br>Alternative Methods of Complia   | substantiated, EASA can approve ance for this AD.   |
|                            | <ol><li>The safety assessment has rec<br/>process and an immediate pub</li></ol>  | quested not to implement the full consultation lication and notification.   |
|                            | <ol> <li>Enquiries regarding this AD sh<br/>Section, Executive Directorate,</li> </ol>  | ould be referred to the Safety Information<br>EASA. E-mail: <u>ADs@easa.europa.eu</u> .                                       |
|                            | 4. For any question concerning th   | ne technical content of the requirements in   |

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