EASA AD No.: 2011-0136R2

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**EASA** 

## **AIRWORTHINESS DIRECTIVE**

AD No.: 2011-0136R2

Date: 27 July 2015

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with EU 748/2012, Part 21.A.3B. In accordance with EU 1321/2014 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EU 1321/2014 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].

Type Approval Holder's Name :		Type/Model designation(s) :
AIRBUS HELICOPTERS		EC 225 helicopters
TCDS Number:	EASA.R.002	
Foreign AD:	Not applicable	
Revision:	This AD revises EASA AD 2011-0136R1 dated 19 July 2011.	
ATA 25	Equipment and Furnis Flotation Gear Protecti	hings – VHF Antenna for Emergency on – Installation
Manufacturer(s):	Airbus Helicopters (formerly Eurocopter)	
Applicability:	EC 225 LP helicopters, all serial numbers, if equipped with an emergency flotation gear and having <u>no</u> antenna installed on the bottom structure, as described in Figure 1 of Alert Service Bulletin (ASB) No EC225-25A086, except helicopters modified in accordance with Airbus Helicopters modification (MOD) 0726840.	
Reason:	A helicopter recently experienced a punctured Right Hand (RH) front float compartment of deployed emergency flotation gear.	
	Eurocopter) showed that the one of the two temperature floats of the front emergence emergency floats, on each	s carried out by Airbus Helicopters (formerly the perforation was caused by an interference with the probes located under helicopter belly near the cry flotation gear. During deployment of the front side of at least one of the LH and RH front float sk of failure to inflate, due to puncturing by the
	This condition, if not correct water in case of emergency	ted, could lead to instability of the helicopter on the y water landing.
		ndition, Airbus Helicopters (formerly Eurocopter) A086 which gives instructions for installation of an

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	inoperative VHF antenna on the RH side of the helicopter, as it has been shown that such installation prevents interference between the RH front float of the emergency flotation gear and the RH temperature probe, thus the stability of the helicopter on the water is ensured if a single float has a punctured compartment.  Prompted by this development, EASA issued AD 2011-0136 to require installation of an inoperative VHF antenna, later revised to reduce the Applicability, to make this AD applicable to helicopters equipped with emergency floatation gear and having no antenna installed on the bottom structure, as described in Figure 1 of Alert Service Bulletin (ASB) No EC225-25A086.  After that AD was issued, Airbus Helicopters developed MOD 0726840. That modification installs protection devices at the level of the temperature probes that prevent interference between the emergency floatation gear front floats and the temperature probes during deployment of the emergency floatation gear and may substitute the inoperative VHF antenna previously installed.  For the reasons described above, this AD is revised to reduce the Applicability and introduce MOD 0726840 as an acceptable alternative method to comply	
Effective Date:	with the requirements of this AD.  Revision 2: 27 July 2015	
	Revision 1: 29 July 2011	
	Original issue: 29 July 2011	
Required Action(s) and Compliance Time(s):	Required as indicated, unless accomplished previously:	
	(1) Within 110 Flight Hours or three (3) months, whichever occurs first after 29 July 2011 [the effective date of this AD at original issue and Revision 1], install an inoperative VHF antenna Part Number 3271 on the RH side of the helicopter, in accordance with the accomplishment instructions of Eurocopter ASB N° EC225-25A086.	
	(2) Modification of a helicopter in accordance with Airbus Helicopters MOD 0726840 and concurrent removal of the inoperative VHF antenna, required to be installed by paragraph (1) of this AD, is an acceptable alternative method to comply with the requirements of paragraph (1) of this AD.	
Ref. Publications:	Airbus Helicopters (formerly Eurocopter) ASB N° EC225-25A086, dated 11 July 2011.	
	The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.	
Remarks :	If requested and appropriately substantiated, EASA can approve     Alternative Methods of Compliance for this AD.	
	<ol> <li>The required actions and the risk allowance have granted the issuance of a Final AD with Request for Comments, postponing the public consultation process after publication.</li> </ol>	
	<ol> <li>Enquiries regarding this AD should be referred to the Safety Information Section, Certification Directorate, EASA. E-mail: <a href="mailto:ADs@easa.europa.eu">ADs@easa.europa.eu</a>.</li> </ol>	
	<ol> <li>For any question concerning the technical content of the requirements in this AD, please contact:         Airbus Helicopters (EBSESB) – Aéroport de Marseille Provence 13725 Marignane Cedex, France;         Telephone +33 (4) 42 85 97 97; Fax +33 (4) 42 85 99 66;         E-mail: Directive.technical-support@airbus.com.</li> </ol>	