



## Airworthiness Directive

**AD No.:** 2025-0279

**Issued:** 10 December 2025

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I Part M.A.301, or Annex Vb Part ML.A.301, as applicable, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I Part M.A.303, or Annex Vb Part ML.A.303, as applicable] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

**Design Approval Holder's Name:**

AIRBUS HELICOPTERS

**Type/Model designation(s):**

H160-B helicopters

**Effective Date:** 24 December 2025

**TCDS Number(s):** EASA.R.516

**Foreign AD:** Not applicable

**Supersedure:** None

### ATA 53 – Fuselage – Cargo Bay – Inspection

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**Manufacturer(s):**

Airbus Helicopters (AH)

**Applicability:**

H160-B helicopters, all serial numbers (s/n), delivered before 21 October 2025 (date of EASA Form 52, or equivalent statement of conformity).

**Definitions:**

For the purpose of this AD, the following definitions apply:

**The ASB:** AH Alert Service Bulletin (ASB) H160-53-40-0001.

**Reason:**

It has been determined that a gap without fire-resistant sealant may be present in the cargo bay near the fuel segregation box on certain helicopters.

This condition, if not detected and corrected, could jeopardize the fire-resistant characteristic of the cargo compartment, potentially resulting, in reduced control of the helicopter in case of fire.



To address this potential unsafe condition, AH published the ASB, requiring an inspection for the presence of the fire-resistant sealant in the frame 4 right side upper corner of the cargo bay, and, depending on findings, application of a bead of fire-resistant sealant.

For the reason described above, this AD requires an inspection and, depending on finding, application of fire-resistant sealant.

#### **Required Action(s) and Compliance Time(s):**

Required as indicated by this AD, unless the action(s) required by this AD have been already accomplished:

#### **Inspection:**

- (1) Within 165 flight hours after the effective date of this AD, inspect the frame 4 right side upper corner of the cargo bay in accordance with the instructions of the ASB.

#### **Corrective Action(s):**

- (2) If, during the inspection as required by paragraph (1) of this AD, it is determined that fire-resistant sealant is missing, as specified in the ASB, before next flight, accomplish the applicable corrective action(s) in accordance with the instructions of the ASB.

#### **Ref. Publications:**

AH ASB H160-53-40-0001 original issue dated 09 December 2025.

The use of later approved revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

#### **Remarks:**

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. Based on the required actions and the compliance time, EASA have decided to issue a Final AD with Request for Comments, postponing the public consultation process until after publication. All interested persons may send their comments, referencing the AD Number, to the E-mail address specified in below Remark 3, prior to 07 January 2026. Only if any comment is received during the consultation period, a Comment Response Document will be published in the [EASA Safety Publications Tool](#), in a compressed ('zipped') file, attached to the record for this AD.
3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu).
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be



installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.

5. For any question concerning the technical content of the requirements in this AD, please contact: Airbus Helicopters (Technical Support) at:  
Web portal: <https://airbusworld.helicopters.airbus.com> / Technical Requests Management, or  
E-mail: [TechnicalSupport.Helicopters@airbus.com](mailto:TechnicalSupport.Helicopters@airbus.com), or Telephone: +33 (0)4 42 859 789.

