COMMONWEALTH OF AUSTRALIA CIVIL AVIATION SAFETY AUTHORITY SCHEDULE OF AIRWORTHINESS DIRECTIVES

Helio Courier Series Aeroplanes

AD/HELIO/1Wing Centre Section Main Spar1/87Amdt 1Carry Thru Assembly

Applicability: All models.

Requirement: 1. For aircraft incorporating wing main spar centre section carry thru assembly P/N 295-030-401, inspect the carry- thru assembly in accordance with Para B) of FAA AD 82-16-08.

- 2. Modify the wing centre section carry-thru assembly P/N 391-030-401 or 295-030-401 as applicable, by installing a full span (wing joint to wing joint) fail safe reinforcement of the lower tube in accordance with a scheme approved by the Secretary. FAA STC's SA 1590CE and SA 1728CE, respectively, are approved schemes.
- 3. After modification in accordance with 2 above, inspect the carry-thru assembly in accordance with Para D) of FAA AD 82-16-08.
- Compliance: For Para 1: At intervals not exceeding 100 hours time in service, until modified in accordance with Para 2.
 - For Para 2: For land planes prior to exceeding 2000 hours time in service.

For float planes or aircraft which have at any time had floats installed prior to exceeding 2000 hours equivalent land plane time in service. To convert float plane time to equivalent land plane time, multiply by two.

- For Para 3: At intervals not exceeding 1500 hours time in service.
- Background: Fatal accidents have occurred overseas as a result of fatigue failures of the wing spar carry-thru assembly lower tube or end fitting. This Directive combines the requirements of the previous AD/HELIO/1 and AD/HELIO/3 Amdt 1 but provides for fail safe reinforcement modification rather than retirement. Most importantly, it eliminates confusion over the "modification" aspect of those earlier AD's; it was not clear that the modification referred to is the joint to joint straps across the full span of the lower tube, rather than the end reinforcement only (ie. the P/N 295-030-401 configuration).