AD/CF6/20 High Pressure Compressor Rotor Stages 11 to 14 Vane to Spool Rubs 11/91

Applicability: CF6-80A and CF6-80C2.

Requirement: Action in accordance with the Technical Requirements of FAA AD 91-15-25 Amendment 39-7090.

Note: GE Service Bulletins 72-459 Revision 2 and 72-130 Revision 2 refer.

Compliance: As specified in the Requirement Document with an effective date of 28 November 1991.

Background: The FAA AD is prompted by reports of HPCR stages 11-14 spool-shafts found in service with spool rub. This condition, if not corrected could result in uncontained engine failure.