AD/CF6/15 High Pressure Compressor Rear Shaft Bolt Hole 8/95 Amdt 2

Applicability: All General Electric CF6-45 and CF6-50 series engines.

Requirement: Action in accordance with the technical requirements of FAA AD 91-10-03 R1 (Amdt. 39-9186).

Note: GE CF6 Service Bulletin 72-958 Rev. 1 also refers.

Compliance: Unless previously accomplished, as specified in the Requirement Document with an effective date of 20 July 1995.

Background: This Directive currently requires an inspection of the HPC Rear Shaft and a revised bolt configuration to reduce the potential for HPC Shaft fractures. HPC Shaft fractures can result in uncontained engine failures.

This amendment reflects the requirements of FAA AD 91-10-03 R1 which adds reference to PMA bolts manufactured by Valley-Todeco.